

The distance between a planet in an elliptical orbit and its star is given by

$$r = a(1 - e \cos E)$$

where a is the semimajor axis, e is eccentricity, and E is the eccentric anomaly.

Eccentric anomaly is a geometric way of describing a planet's position in its orbit. What we care about is the mean anomaly M , which is based on time rather than geometry:

$$M = \frac{2\pi t}{T}$$

where T is the period of the orbit.

The relation between M and E is given by Kepler's equation

$$M = E - e \sin E.$$

The average distance to the star is then

$$\begin{aligned} \frac{1}{T} \int_0^T r \, dt &= \frac{1}{T} \int_0^{2\pi} a(1 - e \cos E) \, dt \\ &= \frac{1}{T} \int_0^{2\pi} a(1 - e \cos E) \, d\left(\frac{TM}{2\pi}\right) \\ &= \frac{1}{T} \int_0^{2\pi} a(1 - e \cos E) \cdot \frac{T}{2\pi} \, d(E - e \sin E) \\ &= \frac{1}{T} \int_0^{2\pi} a(1 - e \cos E) \cdot \frac{T}{2\pi} (1 - e \cos E) \, dE \\ &= \frac{a}{2\pi} \int_0^{2\pi} (1 - e \cos E)^2 \, dE \\ &= a \left(1 + \frac{e^2}{2}\right) \end{aligned}$$